- 1. Course Number and Name: SE 160B: Aerospace Structural Mechanics II
- 2. Credit and Contact Hours: 4 hours of classroom instruction per week.
- 3. Instructor: John B. Kosmatka
- 4. Textbook:
 - Kosmatka, J.B.; <u>Aerospace Structural Mechanics (Course notes for SE-160A, Volume I and II)</u>, UCSD Book Store, 2012.
 - Kosmatka, J.B.; <u>Aerospace Structural Mechanics (Course notes for SE-160B, Volume III)</u>, UCSD Book Store, 2012.
 - Kosmatka, J.B.; Aerospace Structural Mechanics Appendices, UCSD Book Store, 200

5. Specific Course Information:

- **a.** Catalog Description: Analysis of aerospace structures via work-energy principles and finite element analysis. Bending of metallic and laminated composite plates and shells. Static vibration and buckling analysis of simple and built-up aircraft structures. Introduction to wing divergence and flutter, fastener analysis.
- b. Prerequisites: SE 160A. Enrollment restricted to MC25, MC27, and SE27 majors only.
- c. Selected Elective Course

6. Course Objective:

- To teach students methods for analyzing the behavior of metallic and composite plate and shell aerospace structural components.
- To teach students energy-based procedures for analyzing the behavior of thin-wall aerospace structural components.
- To teach students modern computational procedures (finite element) and programs to analyze modern composite stiffened thin-wall aerospace structures.
- To teach students techniques for designing and analyzing structures that require fasteners and/or bonded joints.

7. List of Topics to be Covered:

- Stretching and bending behavior of metallic and composite plates
- Failure of laminated composite plates
- Energy principles in structural analysis (Castigliano's 1st and 2nd theorems)
- Introduction to the finite element method. Formulation of bar, beam, and plate elements
- Static, vibration, and buckling analysis of simple and built-up aerospace structures using a commercial finite element code (MSC NASTRAN)
- Advanced Topics (one or more of the following depending upon available time): structural dynamics, structural stability including buckling and aircraft wing flutter (aeroelasticity), and/or structural optimization
- Analysis for bonded and fastened (bolts, rivets) connections

Person Who Prepared This Description and Date of Preparation

John B. Kosmatka, 3/18/2025